

ASYMMETRY OF PHOTOELECTRON DISTRIBUTION AND ITS DEPENDENCE ON SPACECRAFT POTENTIAL OBTAINED FROM GEOTAIL DATA

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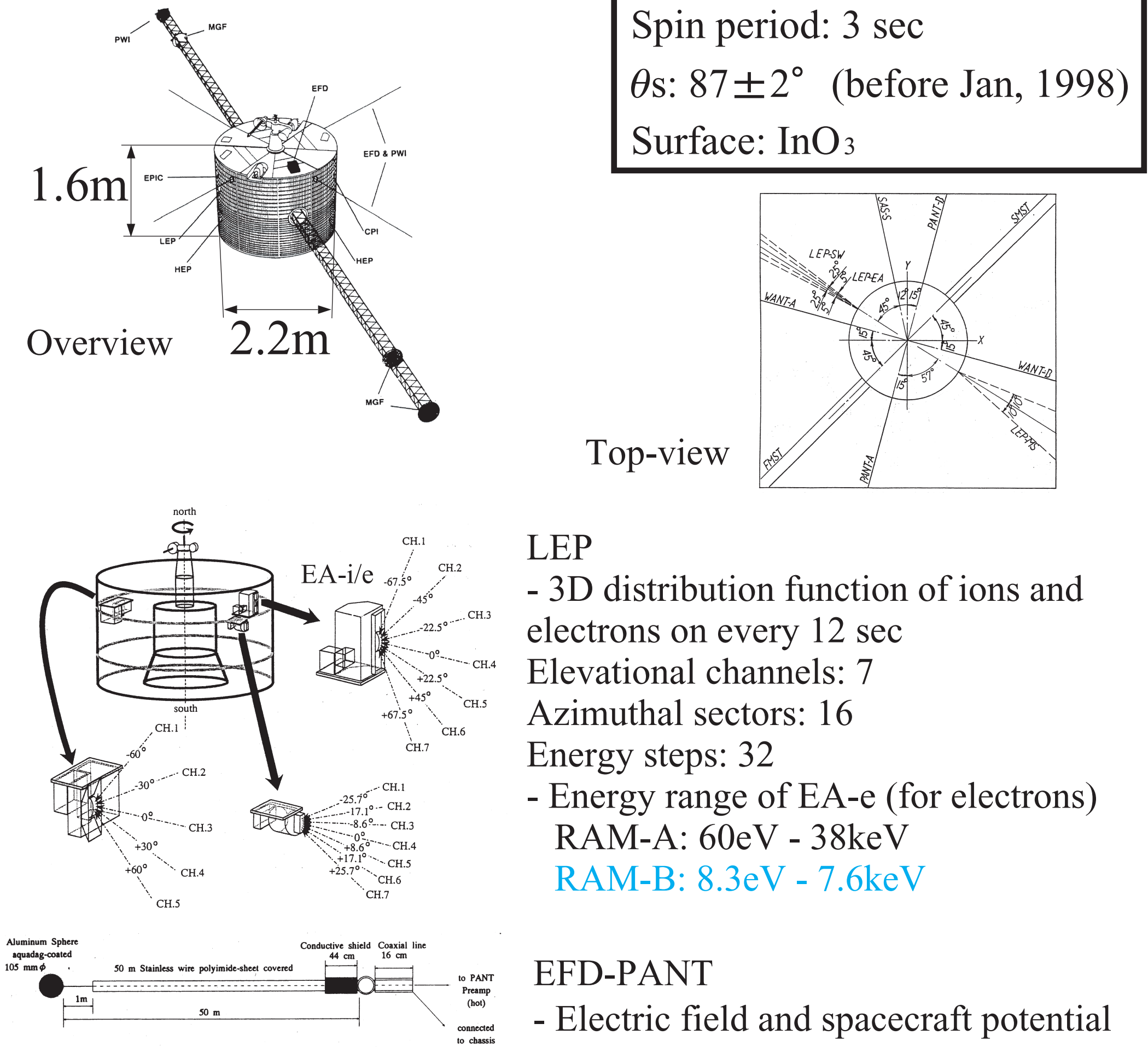
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Abstract

In the Earth's magnetosphere where the spacecraft potential is usually positive, photoelectrons emitted from the spacecraft surface are attracted back to the spacecraft and some of them are detected by electron analyzer onboard. Then photoelectrons are unnecessary contamination. By analyzing such a component detected by LEP/EA-e onboard GEOTAIL spacecraft, we examined velocity/energy distribution functions of photoelectrons, and their relationship to the spacecraft potential.

We found that the ratio of the duskward photoelectron flux to the dawnward flux increases when the spacecraft potential is large, and decreases when it is small. Further analysis revealed, by plotting the ratio as a function of the photoelectron energy normalized by the spacecraft potential (E/V_{sc}), that it is largest when E/V_{sc} is about one third. This result implies the existence of an azimuthal component of electric field as well as the radial component which is regarded to be dominant in an ordinary case around the spacecraft surface, although it is not clear why such electric field is generated around GEOTAIL.

GEOTAIL Spacecraft



Statistical Results

Data selection
- Data from LEP/EA-e worked in RAM-B (low energy) mode & GEOTAIL was sunlit in the magnetosphere

Criteria for GEOTAIL being out of the magnetosphere
(In the magnetosheath or the solar wind)

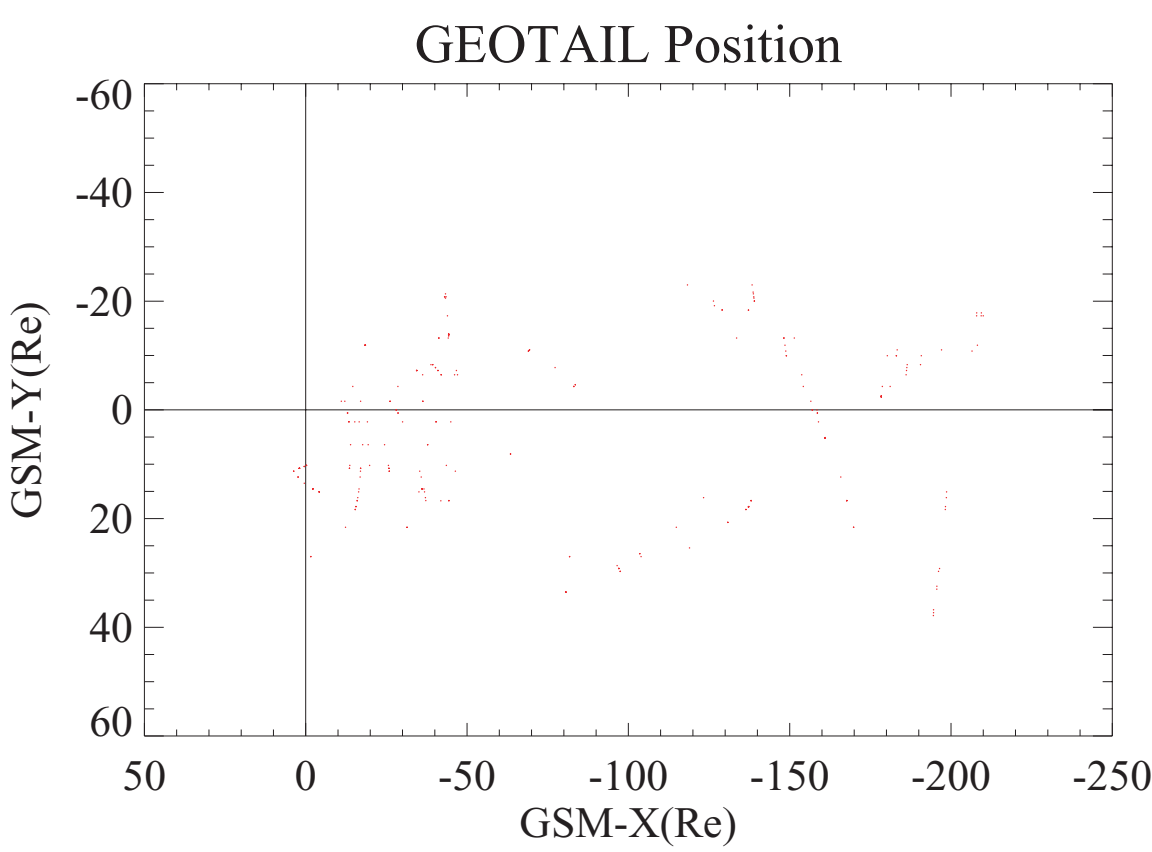
- $N_i > 0.5 \text{ cm}^{-3}$
- $V_{ix} < -200 \text{ km/s}$
- $T_i < 400 \text{ eV}$

and

- $P_d = M_i N_i V_{ix}^2 > 0.5 \text{ nPa}$

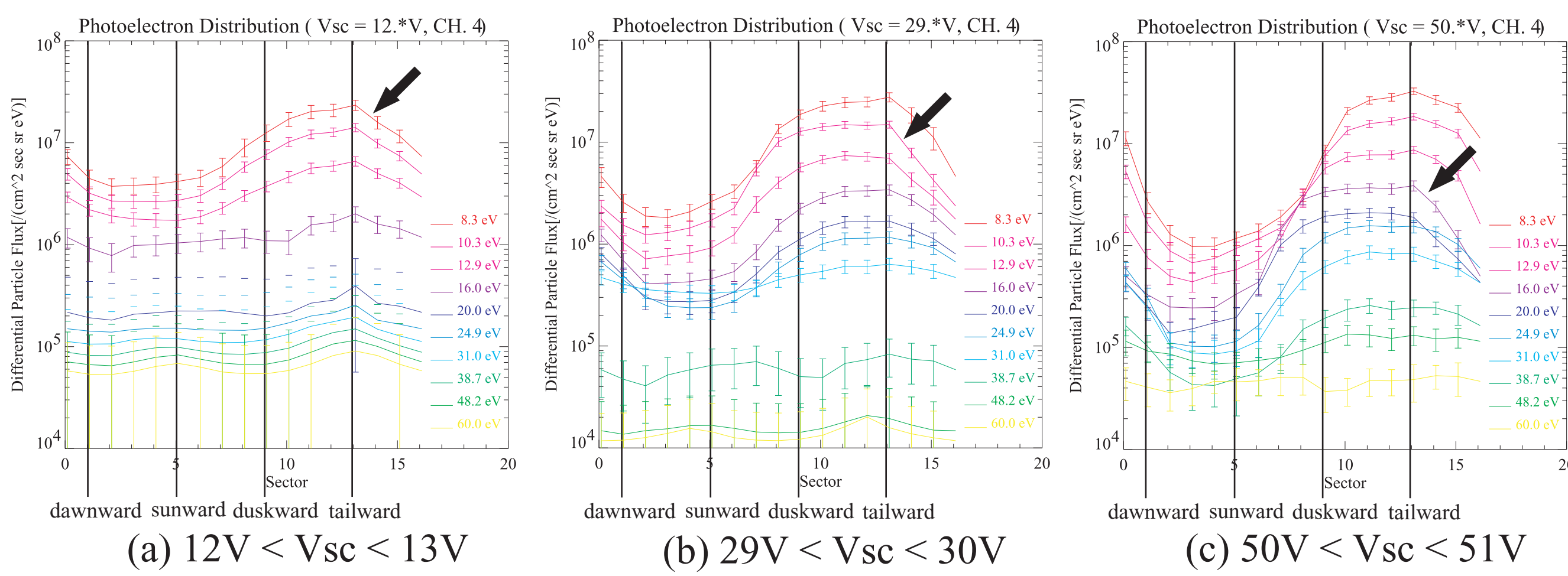
→ 168,166 samplings

[Nakamura et al., 1997] (with some modification)



Data sorting
- by spacecraft potential with one-volt bin

Averaged 2-D differential flux of electrons of each V_{sc} -bin (using only Ch. 4 data of LEP/EA-e)



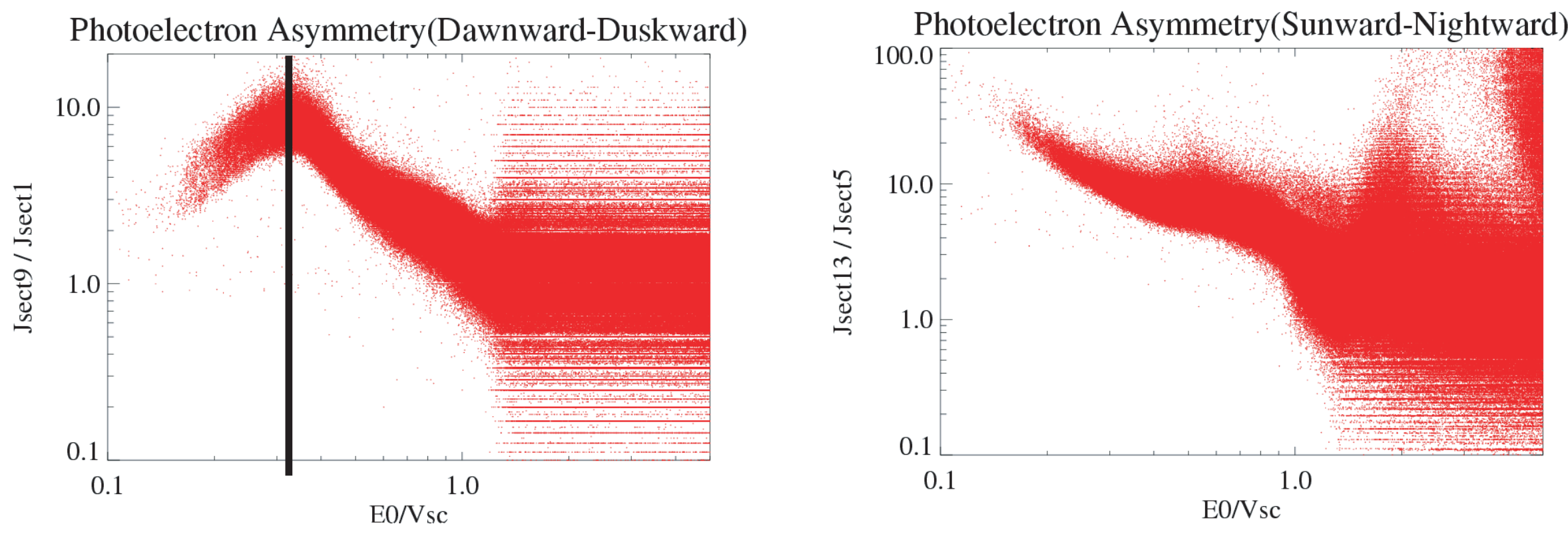
Strong dawn-dusk asymmetry is seen at (a) 8.3eV, (b) 10.3eV, and (c) 16.0eV.

Lowest energy step

$V_{sc}/3$

→ Does photoelectron asymmetry depend on Energy/ V_{sc} rather than V_{sc} itself?

Relation between photoelectron distribution and Energy/ V_{sc}



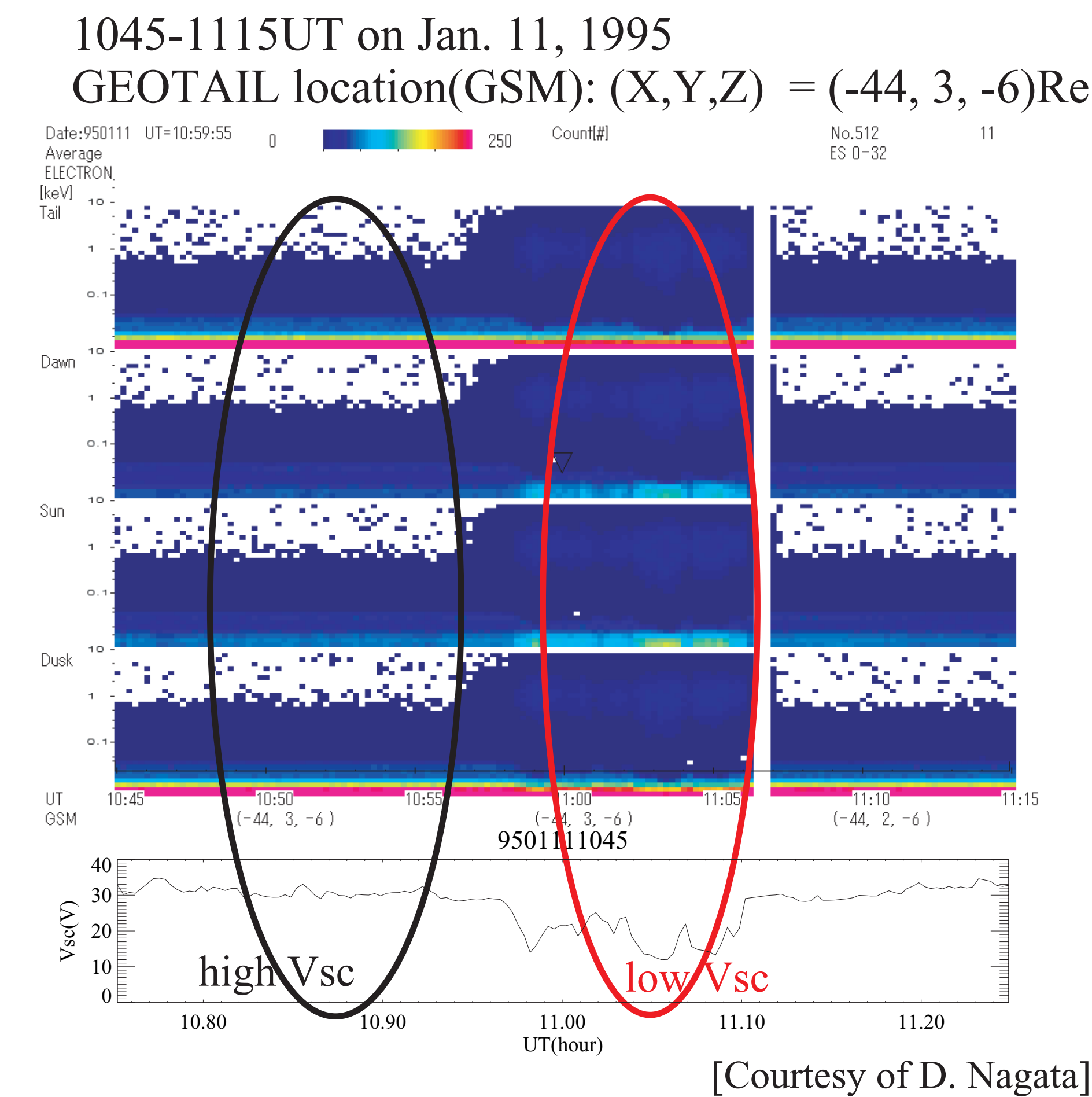
The ratio of duskward photoelec. flux to dawnward photoelec. flux

The ratio of tailward photoelec. flux to sunward photoelec. flux

both as a function of Energy/ V_{sc}

→ Strong dependence of photoelectron asymmetry on Energy/ V_{sc}

Event Study

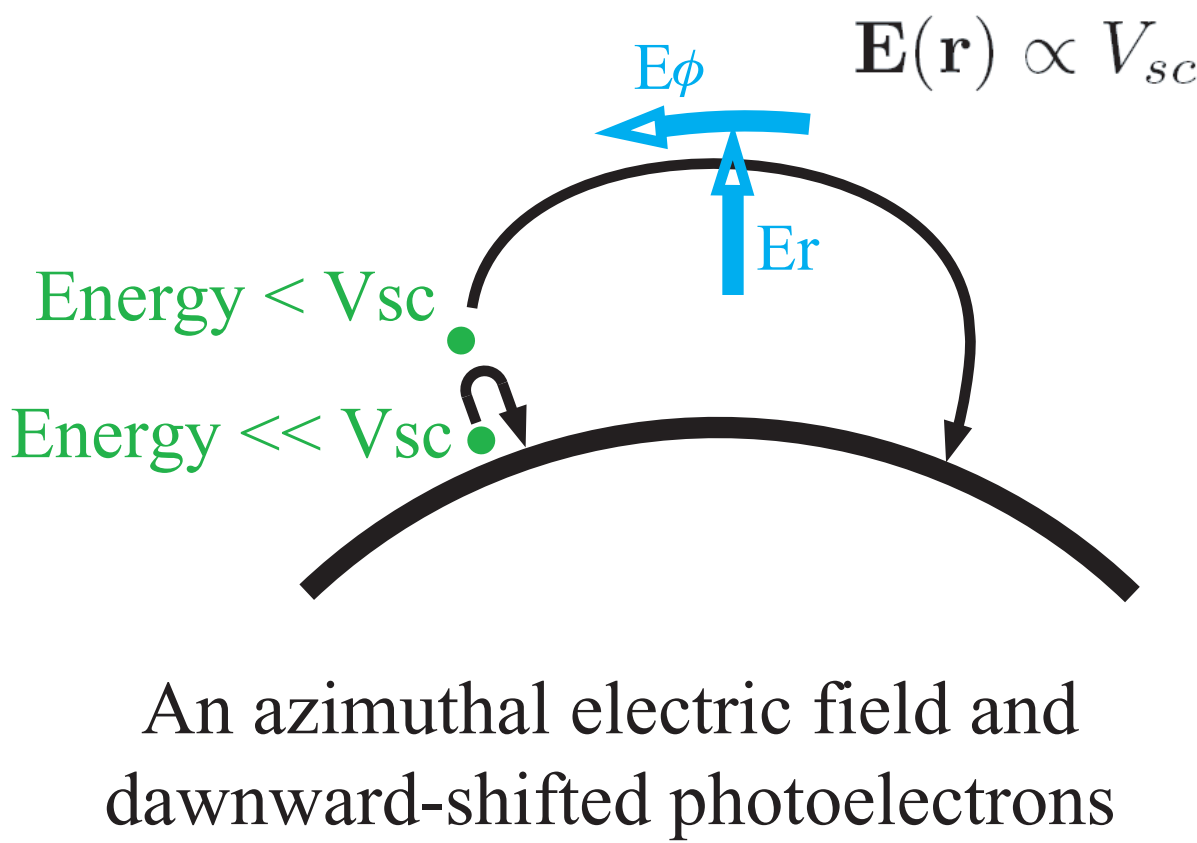


- Dawnward photoelec. < Duskward photoelec.
 - Dawnward and sunward photoelec. counting rates decrease as V_{sc} increases.
 - Duskward and tailward photoelec. counting rates increase as V_{sc} increases.
- Dependence of photoelectron asymmetry on V_{sc}

Discussions

A possible reason for the asymmetry

Azimuthal electric field generated by spacecraft charging



Azimuthal electric field of spacecraft origin is the most probable reason.

However, a couple of masts are deployed symmetrically with respect to FOV of LEP/EA-e.

→ How to explain the source of such azimuthal electric field ??

Other possibilities

Gyromotion of electrons due to the ambient magnetic field

Electron gyroradius:
 $r_c = 3.4 \cdot 10^3 \sqrt{K/B}$
(K : elec. energy, B : magnetic field)
For $K = 1 \text{ eV}$ and $B \approx 30 \text{ nT}$ (in lobe)
 $r_c \approx 110 \text{ m} \gg \text{Spacecraft Size}$
→ Not effective

Effect of ambient electric field (dawn-dusk direction)

Ambient $E_y \sim 0.1 - 1 \text{ mV/m}$
Comparison with E_r
 $E_r \approx V_{sc}/\lambda_D \sim V_{sc}/\lambda_{Dph}$
cf. Debye length of photoelec.
 $\lambda_{Dph} \sim 4.7 \text{ m}$ ($6 \text{ V} < V_{sc} < 25 \text{ V}$)
[Nakagawa et al., 2000]
Debye length of ambient elec.
 $\lambda_{Damb} \sim 100 - 1000 \text{ m}$
[Baumjohann and Treumann, 1997]
 $E_r \sim 1 \text{ V/m} \gg E_y$
→ Not effective

Summary and Conclusion

We examined a 2-D photoelectron distribution around GEOTAIL and its dependence on spacecraft potential.

Dawn-dusk asymmetry as well as day-night asymmetry was found. They strongly depend on the ratio of photoelectron energy to spacecraft potential.

Such dependence suggests the existence of the azimuthal electric field of spacecraft origin, although the source of such an electric field has not been identified.